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# SHAPE MEMORY AND PSEUDOELASTIC BEHAVIOR OF 51.5%Ni–Ti SINGLE CRYSTALS IN SOLUTIONIZED AND OVERAGED STATE

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Abstract—Deformation of nickel rich (51.5%Ni) Ni-Ti single crystals are investigated over a wide range of temperatures (77-440 K) and strain levels in compression as high as 9%. These alloys combine high strength with an unusually wide pseudoleasticity temperature interval (near 200 K) and can be exploited to suit specific applications. The slip deformation in [001] orientation can not occur due to the prevailing slip systems, as confirmed by transmission electron microscopy. Consequently, the [001] orientation exhibited pseudoleastic deformation at temperatures ranging from 77 to 283 K for the solutionized case and 273-440 K for the aged condition respectively. The critical transformation stress levels were in the range 800-1800 MPa for the solutionized case, and 200-1000 MPa for the aged case depending on the temperature and specimen orientation. These stress levels are considerably higher compared to the near equiatomic Ni compositions of these class of alloys. On the other hand, the maximum transformation strains, measured from incremental straining experiments in compression, were lower compared to both the phenomenological theory with Type II twinning and the previous experimental work on 50.8% Ni NiTi crystals. A new theory for compound twinning is introduced with lattice invariant shear as a solution, and relies on the successive austenite phase (B2) to intermediate phase (R) to martensite phase (B 19') transformation. The compound twinning model predicts lower transformation strains compared to the Type II twinning case lending an explanation of the experimental transformation strain levels. © 2001 Acta Materialia Inc. Published by Elsevier Science Ltd. All rights reserved.

- *Keywords:* Shape memory; Martensite; Single crystal; Transmission electron microscopy (TEM); Phase trans formations
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# 1. INTRODUCTION

Phase transformations in nickel titanium alloys have 71 been studied extensively with most emphasis placed 72 on near equiatomic Ni-Ti compositions [1]. Two 73 types of material response have been examined, 74 namely "shape memory" and "pseudoelasticity". The 75 "shape memory" refers to the transformation of the 76 material from martensite to austenite upon heating to 77 a temperature exceeding the austenite start tempera-78 ture. "Pseudoelasticity" is the forward transformation 79 upon loading and reverse transformation upon 80 unloading at temperatures above the austenite finish 81 temperature. The pseudoleasticity temperature inter-82 val is approximately 80 K in equiatomic NiTi alloys. 83

† To whom all correspondence should be addressed. *E-mail address:* huseyin@uiuc.edu (H. Sehitoglu) This temperature range is curtailed by the lack of slip 84 resistance near the  $M_d$  temperature. The  $M_d$  tempera-85 ture is defined as the highest temperature where 86 pseudoleasticity can be observed. Majority of the 87 work on NiTi alloys has considered aged conditions 88 because the aging treatments produce pseudoleasticity 89 near room temperature [1–3] and also raise the  $M_d$ 90 temperature. It has also been known that increasing 91 the austenite strength increases slip resistance thereby 92 extending the utility of these alloys to a wider tem-93 perature range. The strengthening of the austenite is 94 achieved via deformation processing routes or with 95 the fine precipitate structure. Nickel rich compo-96 sitions (with at. Ni exceeding 51%) of NiTi alloys can be used to strengthen the austenite domains via high volume fraction of precipitates. Despite their strong potential, nickel rich NiTi alloys have been 100 considered in only few studies [4]. The only notable 101

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work has been undertaken by Miyazaki and col-102 leagues [4] who studied polycrystalline 51.6% Ni at. 103 NiTi alloys. They noted that the 51.6% at. Ni NiTi 104 alloy exhibits a lower martensite start temperature and 105 higher slip resistance compared to its lower Ni 106 counterparts. Further work is needed in Ni rich NiTi 107 alloys to examine the possibility of a wider pseudole-108 asticity temperature interval and also high strength in 109 both the austenitic and martensitic phases. 110

Upon austenite to martensite transformation in 111 NiTi alloys the martensite is composed of internal 112 twins in the B19' phase, i.e., a twin related Corre-113 spondence Variant Pair (CVP) forms [2]. This mar-114 tensite variant can be produced either by a single step 115 of B2 $\rightarrow$ B19' transformation [5, 6], or by the success-116 ive B2 $\rightarrow$ R-Phase $\rightarrow$ B19' transformation [7–9]. Type 117 II twinning has been dominantly observed in experi-118 ments and is confirmed to be lattice invariant shear. 119 For the case of aged alloys "compound twinning" has 120 also been identified experimentally [10-12] but ruled 121 out as a lattice invariant shear based on theoretical 122 considerations [13]. In other words, a theoretical sol-123 ution for the twin and habit planes could not be 124 obtained. An original calculation is presented in this 125 paper addressing the successive B2 to R to B19' 126 transformation that produces "compound twinning" 127 as a lattice invariant shear solution. The transform-128 ation strains associated with the compound twinning 129 are shown to be lower compared to the Type II twin-130 ning solutions lending an explanation to the exper-131 imentally observed transformation strains levels. 132

Our previous work [2, 3] focussed on 50.8% at. Ni 133 single crystals, and in this work the emphasis is 134 placed on the 51.5% at. Ni NiTi composition. Also, 135 in previous work we considered only aged materials, 136 in the current work [2] we study both aged and sol-137 utionized materials. Higher strength levels for both 138 the B2 (austenite) and the B19' (martensite) phases 139 are expected in this alloy. We demonstrate that the 140 behavior of this class of alloys is similar to the 50.8% 141 at. Ni case but with greater potential in actuator appli-142 cations where ultra high strength in compression is 143 desired. On the other hand, the transformation strains 144 145 in the case of 51.5% at. Ni alloys (both in the solutionized and the overaged cases) are lower compared 146 to the 50.8% at. Ni NiTi alloys (in compression). 147 Possible reasons for this difference are rationalized 148 based on a new theory for austenite to martensite 149 transformation discussed above which admits the R-150 phase as an intermediate structure [5-9]. 151

We note that in previous treatments the defor-152 mation experiments were conducted on polycrystals 153 [1]. In the present work, five different crystallo-154 graphic orientations ([001], [110], [111], [122] and 155 [012] in the austenite crystal frame) were studied 156 experimentally. These orientations were chosen 157 because [001] and [012] represent the most favorable 158 159 orientations for transformation in compression while the [111] and [122] represent the most difficult orien-160 tations for transformation in compression. Further-161

more, the [001] case is of special interest because 162 plastic deformation in this orientation is curtailed due 163 to the operating  $\{001\} < 001 >$  and  $\{011\} < 001 >$  slip 164 systems. TEM photographs of deformed [001] speci-165 mens confirm the lack of slip activity in this case. 166 These results clearly show that the [001] crystal 167 orientation exhibits an increase in flow stress with 168 increasing temperature beyond the traditional pseudo-169 elasticity regime for both the solutionized and over-170 aged materials. Since the experiments were conducted 171 at multiple temperature levels, it was possible to 172 determine precisely the temperature range over which 173 pseudoelastic response occurred. For the solutionized 174 case, the  $M_s$  temperature is below 77 K and pseudo-175 elasticity has been observed in the experiments in the 176 range 77–283 K. For the case of aged alloys, the  $M_s$ 177 temperature is near 236 K and pseudoelasticity is 178 observed in the range 273-440 K. 179

The tensile ductility of the 51.5% at. Ni alloys is 180 decreased compared to 50.8% Ni alloys. Because the 181 present experiments were conducted under com-182 pression, fracture of the specimens did not occur. This 183 is an important advantage of utilizing compression 184 specimens. Consequently, the experiments have been 185 carried out to strains of 9% or higher without diffi-186 culties, allowing examination of the plastic defor-187 mation behavior of austenite and martensite phases. 188

# 2. EXPERIMENTAL PROCEDURE

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Single crystal Ti-51.5% Ni samples were prepared from a single material batch. The samples were grown 191 by the Bridgman technique in an inert gas atmos-192 phere. The orientation of single crystal specimens was 193 determined by using Laue back-scatter diffraction 194 patterns. Solutionizing of the specimens was conduc-195 ted at 1273 K for 2 h in an inert gas atmosphere. 196 Then, one set of specimens was examined in the sol-197 utionized state while the other set of specimens was 198 aged at 823 K for 1.5 h. A Perkins-Elmer differential 199 scanning calorimeter (DSC) was used to determine 200 the transformation temperatures for the overaged 201 case. The DSC results are shown in Fig. 1(a). The 202 austenite start temperature,  $A_s$ , for overaged samples 203 is 284 K while the  $M_s$ , martensite start temperature, 204 is approximately 236 K. Other transformation tem-205 peratures are  $A_f = 300$  K and  $M_f = 223$  K. An inter-206 mediate phase, called the R-phase, appears in the 207 DSC results which transforms to martensite upon 208 further cooling. For the solutionized case, the trans-209 formation temperatures were too low (<77 K) to be 210 detected by the DSC technique. 211

We note that the overaged microstructure is shown 212 in Fig. 1(b) and (c). The approximate size of the 213 Ti<sub>3</sub>Ni<sub>4</sub> precipitates is 750 nm and the volume fraction 214 is nearly 20%. Four variants of the Ti<sub>3</sub>Ni<sub>4</sub> precipitates 215 are present but fewer than four are visible on the 216 microphotographs due to the specimen orientation 217 [10]. The precipitates have a lens-like shape with their 218 normal parallel to the [111]<sub>B2</sub> orientation of the 219

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Fig. 1. (a) The differential scanning calorimetry results for the 51.5%Ni–NiTi alloy (overaged 823 K 1.5 h case). (b) The four variant precipitate structure in 51.5%Ni–Ti (only three variants are visible in this case) showing the undeformed structure. (c) The dislocation structure between the precipitates for the [111] overaged case after incremental straining deformation.

matrix. In Fig. 1(b) the precipitate structure is shown 220 for the undeformed state of the material. We note that 221 the precipitates do not transform to martensite. A sig-222 nificant decrease in the amount of material that can 223 undergo transformation means that the experimental 224 transformation strains would be lower than the theor-225 etical values. In Fig. 1(c) the microphotograph of the 226 [111] specimen is shown which has been deformed 227 under incremental straining conditions. As discussed 228 229 later, this orientation is favorable for slip deformation with presence of dislocations along the austenite 230 channels (Fig. 1(c)). 231

The specimen has a square crossection with 4 mm 232 width and 8 mm height. In the experiments, the loads 233 were measured with a load cell, and strains were mea-234 sured with a miniature MTS (Materials Test Systems) 235 extensometer with a 3 mm gage length. The use of the 236 miniature extensometer circumvents the end effects 237 associated with stroke measurements. The strain rates 238 were maintained at  $10^{-4}$  l/s to minimize both the rate 239 effects and temperature rise during the experiments. 240 The critical stress for transformation as a function 241 of temperature has been determined with a series of 242 243 deformation experiments. The results are summarized in Figs 2 and 3 for solutionized and overaged cases 244 245 respectively. Companion samples were used to establish these points to avoid previous deformation effects 246 on the results. The  $M_s$  is obtained by extrapolation of 247 the stress-temperature relation to zero stress as 230 248 K (average of all orientations) for the overaged case. 249 This is consistent with the  $M_s$  value obtained from 250 DSC measurements. Such a comparison is not mean-251 ingful in the solutionized case and the behavior is 252 rather complex. 253

200 nm

At low temperatures, the strength of the sol-254 utionized case is rather high for orientations 255 (especially that do [111] not favor the 256 transformation). The [001] orientation for both the 257 aged and solutionized cases exhibits similar stress-258 temperature behavior except that the  $M_s$  temperatures 259 are drastically different (near 230 K for aged and <77 260 K for the solutionized cases respectively). The slope 261 of the stress-temperature relation in the pseudoleas-262 ticity regime is linear and this region is described with 263 the Clausius-Clapeyron relation. The slope of the 264 Clausius–Clapeyron curve  $(A \rightarrow M \text{ region})$  is on the 265 average 6.5 MPa/°C. The  $M_d$  temperature is defined 266 here as the maximum temperature (nearly 360 K for 267 the overaged case) below which pseudoleastic defor-268 mation occurs. Note that the  $M_{\rm d}$  temperature depends 269 on the crystal orientation but it cannot be identified 270 for the [001] case because slip is curtailed in this 271



Fig. 2. Transformation stress (0.2% offset) as a function of temperature under compression; solutionized case.



Fig. 3. Transformation stress (0.2% offset) as a function of temperature under compression; overaged case.

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orientation. Also, for the case of solutionized samples, the  $M_d$  cannot be clearly established for any of the crystal orientations.

The most significant finding in these experiments is the following. In the [001] orientation the critical stress for transformation continues to increase with temperatures well past 400 K for both the overaged and the solutionized cases. TEM photographs (to be shown later) indeed confirm the lack of slip in this orientation.

# 3. CALCULATION OF THEORETICAL TRANSFORMATION STRAINS

Upon austenite to martensite transformation in 284 NiTi alloys (either due to stressing above  $M_s$ , or coo-285 ling below  $M_s$ ) the martensite is composed of internal 286 twins in the B19' phase, i.e., a twin related Corre-287 spondence Variant Pair (CVP) forms [2]. This mar-288 289 tensite variant can be produced either by a single step of B2 $\rightarrow$ B19' transformation [5–7], or by the success-290 ive B2 $\rightarrow$ R-Phase $\rightarrow$ B19' transformation [7–9]. There 291

are 12 lattice correspondence variants for  $B2 \rightarrow B19'$ 292 transformation [6] and 8 twinning types can be for-293 med within these 12 lattice correspondence variants 294 [2, 14]. Based on crystallography, some twinned vari-295 ant pairs are able to form the so called "habit plane" 296 variants since their deformations are compatible with 297 B2 phase or R-Phase. Type II twinning has been 298 dominantly observed in experiments and is confirmed 299 to be lattice invariant shear. In Type II twinning the 300 twin plane normal is irrational. For the case of aged 301 alloys compound twinning has also been identified 302 experimentally [9-12] but ruled out as a lattice 303 invariant shear based on theoretical considerations 304 [13]. 305

In the present work, regardless of the twin type, 306 each martensite CVP is described by a unique habit 307 plane normal, m, and transformation direction, b. 308 Two cases are considered in this work: (i) the Type 309 II twinning solution, and (ii) the two step transform-310  $B2 \rightarrow R-Phase \rightarrow B19'$ ations of transformation 311 resulting in compound twinning. We now summarize 312 the methods for determination of the habit plane para-313

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meters and the transformation strains associated witheach solution.

#### 316 3.1. Type II twinning

In the parent phase coordinate system (B2), the 317 deformation matrices of the 12 B19' lattice corre-318 spondence variants are designated as  $U_1, U_2, ..., U_{12}$ . 319 For a given variant pair (Ui, Uj), the twin plane n320 and twin shear *a* can be determined by using the con-321 dition that the interface between the two lattice corre-322 spondence variants in the twin is an invariant plane 323 (unrotated and undistorted) [14, 15] 324

 $\mathbf{R}_{ii}\mathbf{U}_i - U_i = \mathbf{a} \otimes \mathbf{n}$ 

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where **Rij** is an orthogonal tensor (of rank 2) 328 satisfying  $\mathbf{R}_{ij}^T \mathbf{R}_{ij} = \mathbf{I}$  ( $\mathbf{I}$  is second rank identity tensor, 329 and the superscript T represents the transpose of a 330 matrix). The  $\otimes$  represents a dyadic product. The ten-331 sor **Rij** represents the relative rotation between the 332 two variants. The twinned martensite is composed of 333 correspondence variant pairs with a certain volume 334 ratio. When there are finite number of twin layers, the 335 deformation of martensite is represented as 336

$$\boldsymbol{F}_{\mathbf{M}} = \boldsymbol{R}_h[f\boldsymbol{R}_{ij}\boldsymbol{U}_j + (1-f)\boldsymbol{U}_i]$$

where Ui and Uj are the two lattice correspondence variants in the twin, and (1-f) and f are respectively their volume fractions. The tensor Rh, is the relative rotation between the twinned martensite and the parent phase. The habit plane m and transformation shear b can be obtained by

 $F_{\rm M}-I=b\otimes m$ 

where I is the identity tensor representing the unde-349 formed austenite, m is the habit plane normal and b350 is the shear of the martensite. In equations (1)–(3), 351 the known parameters are Ui, Uj, and all the other 352 unknowns can be solved from the equations. The lat-353 tice parameters of B2 phase and B19' phases must be 354 known [6]. For the B2 phase  $a_0 = 3.015$  Å, and for 355 the B19' phase: a = 2.889 Å, b = 4.120 Å, 356  $c = 4.622 \text{ Å}, \theta = 96.8^{\circ}.$ 357

The base crystallographic parameters for sol-358 utionized NiTi produce two solutions as: m =359  $0.8889, 0.4044, 0.2152, \ b = < 0.0568, 0.0637, 0.0991 >$ 360 and m = 0.3762, 0.5136, 0.7712, b = < 0.1195, 0.0485,361 0.0216 > [2]. The habit plane solutions above are 362 obtained using {0.7205, 1, 0}<001> Type II-1 twin-363 ning (we use the term Type II-1 because there exists 364 another {1, 1, 3.0495}<-1, 1, 0> Type II variants 365 which are designated as Type II-2) which is consist-366 367 ent with the dominant mode experimentally observed. In Type II twinning the twin plane normal is irrational 368 while in Type I the twin plane normal is rational. 369

Once the habit plane normals and transformation 370 shears are determined, it is possible to establish the 371 transformation strain as, 372

$$\varepsilon = \frac{1}{2} (F_{\mathbf{M}}^T \cdot F_{\mathbf{M}} - I) \tag{4}$$

$$=\frac{1}{2}[\mathbf{b}\otimes\mathbf{m}+\mathbf{m}\otimes\mathbf{b}+(\mathbf{b}\cdot\mathbf{b})\mathbf{m}\otimes\mathbf{m}]$$
<sup>374</sup>
<sup>375</sup>

The last term in equation (4) represents finite strain 377 effects. The transformation strain contours corre-378 sponding to equation (4) are provided in Fig. 4(a) for 379 compression and Fig. 4(b) for tension. The small 380 deformation theory (without the last term in equation 381 (4)) produces recoverable strains in compression 382 which are in approximately 10% higher than the large 383 deformation theory (Green strain) used in equation 384 (4). The transformation strain results for compression 385 for the five orientations considered in this study are 386 given in Table 1 as the third column. The resolved 387 shear stress factor (RSSF) for Type II-1 twinning 388 (compression) is included in Table 1 as second col-389 umn. The first column in Table 1 lists all the orien-390 tations of interest in our study. The resolved shear 391 stress factor, RSSF (for Type II-1 twinning), is simply 392 defined as, 393

$$RSSF = (\boldsymbol{b} \cdot \boldsymbol{e})(\boldsymbol{m} \cdot \boldsymbol{e})/|\boldsymbol{b}| \qquad (5) \quad {}_{394}$$

where e denotes the crystallographic direction of <sup>397</sup> interest. Note that the RSSF can be larger than 0.5 <sup>398</sup> because the transformation direction and the transformation plane are not necessarily orthogonal. <sup>400</sup>

# 3.2. Compound twinning 401

It has been generally accepted that [011] Type II 402 and  $[11\overline{1}]$  Type I are the major twinning modes that 403 produce lattice invariant shear in B2 to B19' trans-404 formation. However, there is increasing evidence that 405 [001] compound twinning mode has been observed 406 repeatedly by several researchers [10-12]. In NiTi 407 alloys there exists a so-called "premartensitic" 408 rhombohedral phase (R-Phase) formed as a precursor 409 to the B19' martensitic transformation [7-9]. Based 410 on the phenomenological theory of crystallographic 411 transformation or the energy minimization theory 412 [15], it is impossible for the compound twin to consti-413 tute a lattice invariant shear during the B2 to B19' 414 martensitic transformation. In view of the experi-415 mental evidence for the observation of compound 416 twinning in B19' martensite [10-12, 16, 17], in this 417 work, we analyzed a compound twinned B19' mar-418 tensite during the R-Phase to B19' transformation. 419

There are four lattice correspondence variants for<br/>the B2 to R-Phase transition and the corresponding<br/>deformations can be represented in the B2 coordinate<br/>system as [8, 9]420<br/>421

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Fig. 4. (a) Transformation strain contours for Type II twinning case [2] (compression). (b) Transformation strain contours for Type II twinning case (tension). (c) A schematic of two stage transformation from the austenite to R-Phase to martensite. (d) The transformation strain contours for B2 to R-Phase to compound twinned B19' (compression). (e) The transformation strain contours for B2 to R-Phase to compound twinned B19' (tension).

Table 1. Theoretical RSSF (Resolved Shear Stress Factor) and the compressive transformation strains for the Type-II-1 twinning, compound twinning of B19', and the deformed austenite Type-II-1 case for compression. The last two columns are the experimentally determined recoverable strains for the solutionized and the overaged cases respectively (compression)

Crystallographic Direction	RSSF B2—B19' -Type II-1 Twinning	Type II-1 Twinning- (%)	B2→R→B19' Strain of Compound Twinned B19' (%)	Experimental Recoverable Strains (%)	
				Solutionized	Overaged
[001]	0.386	4.38	4.73	3.9	3.3
[110]	0.430	5.06	3.57	4.2	3.7
[111]	0.253	2.98	3.79	1.2ª	3.0
[012]	0.508	6.23	4.99	3.7	3.2
[122]	0.370	4.54	4.24	$0.76^{a}$	2.3

<sup>a</sup> plastic deformation dominates the stress-strain response.

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$$U_{1} = \begin{bmatrix} \alpha & \beta & \beta \\ \beta & \alpha & \beta \\ \beta & \beta & \alpha \end{bmatrix}, U_{2} = \begin{bmatrix} \alpha & -\beta & \beta \\ -\beta & \alpha & -\beta \\ \beta & -\beta & \alpha \end{bmatrix}, \quad (6)$$

$$U_{3} = \begin{bmatrix} \alpha & \beta & -\beta \\ \beta & \alpha & -\beta \\ -\beta & -\beta & \alpha \end{bmatrix}, U_{4} = \begin{bmatrix} \alpha & -\beta & -\beta \\ -\beta & \alpha & \beta \\ -\beta & \beta & \alpha \end{bmatrix} (7)$$

429 where  $\alpha = (\sqrt{1 + 2\cos\psi} + 2\sqrt{1 - \cos\psi})/3$ ,  $\beta = (\sqrt{1 + 2\cos\psi})/3$ 430  $-\sqrt{1 - \cos\psi}/3$  and  $\psi$  is the rhombohedral angle. The 431 B2 phase and rhombohedral phase have the same unit 432 cell length of  $a_0 = 3.015$  Å.

433 Similar to the B2 to B19' transformation, B2 to R434 Phase transformation is controlled by

$$T_{AB}U_B - U_A = a_B \otimes n_B \tag{8}$$

$$F_{\mathbf{R}} = T_h [f_{\mathbf{R}} T_{\mathbf{A}\mathbf{B}} U_{\mathbf{B}} + (1 - f_{\mathbf{R}}) U_{\mathbf{A}}]$$
(9) 439

$$F_{\rm R} - I = \boldsymbol{b}_{\rm R} \otimes \boldsymbol{m}_{\rm R} \tag{10} \quad 441$$

where  $U_A$ ,  $U_B$  are two variants within the 4 R-Phase variants,  $T_{AB}$  is the relative rotation between variants A and B in the twin,  $(1-f_R)$  and  $f_R$  are the volume fraction of variants A and B, respectively. The tensor  $F_{\rm R}$  represents the average deformation of twinned R-Phase, *I* is the deformation of austenite, *Th* is the relative rotation between R-Phase and austenite, and  $a_{\rm R}$ ,  $n_{\rm R}$ ,  $b_{\rm R}$ ,  $m_{\rm R}$  are respectively the twinning shear, twinning plane, transformation shear and habit plane of R-Phase. The relevant vectors are shown in the schematic given as Fig. 4(c). From these equations 

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we can see that the R-Phase is not a single lattice 455 correspondence variant, but is twin related. This has 456 also been confirmed with experimental observations 457 [8, 9] Two solutions of  $\{100\} < 011 >$ 458 and  $\{110\} < 011 >$  compound twins are obtained from 459 equation (8) depending on the rhombohedral angle 460  $\psi$ . The volume fraction  $f_{\rm R}$ , habit plane normal  $m_{\rm R}$  and 461 transformation shear  $b_{\mathbf{R}}$  can be obtained from equa-462 tions (9) and (10) for a given variant pair  $U_{\rm A}$  and  $U_{\rm B}$ . 463 There are 24 habit plane variants with {100}<011> 464 compound twinning when  $\psi > 90^\circ$ , while no variants 465 with {100}<011> compound twinning exist when 466  $\psi < 90^{\circ}$ . The transformation shears and habit plane 467 normals for the  $\{100\} < 011 >$  compound twinning 468 variants are  $\psi$ -dependent, as well as the volume frac-469 tion  $f_{\rm R}$ . The habit plane variants with  $\{110\} < 001 >$ 470 compound twins can be obtained in both cases of 471  $\psi > 90^{\circ}$  and  $\psi < 90^{\circ}$ , and the number of variants is 12. 472 The transformation shears and habit plane normals for 473 these variants are  $\psi$ -dependent, while the volume 474 fraction  $f_{\rm R}$  (=0.5) is  $\psi$ -independent. 475

The R-Phase to B19' martensitic transformation iscontrolled by

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$$F_{\mathbf{B}\mathbf{1}\mathbf{9}'} - F_{\mathbf{R}} = \boldsymbol{b}_{\mathbf{B}\mathbf{1}\mathbf{9}'} \otimes \boldsymbol{m}_{\mathbf{B}\mathbf{1}\mathbf{9}'} \tag{11}$$

where  $m_{B19'}$  is the habit plane normal between R-481 Phase and B19',  $b_{B19'}$  is the transformation shear. The 482  $F_{B19'}$  is the average deformation of B19' martensite, 483 which has the same form as  $F_{\rm M}$  in equation (2). Sel-484 ecting a certain  $F_{\mathbf{R}}$  as matrix and substituting equa-485 tions (1) and (2) into equation (11),  $b_{B19'}$  and  $m_{B19'}$ 486 can be obtained. When  $|\psi - 90^{\circ}| \le 2.7^{\circ}$ , the results are 487 similar with those of single step B2 to B19' trans-488 formation, specifically, there are 48 habit plane vari-489 ants with Type I-1, Type I-2, Type II-1 and Type II-490 2 twins [2]. When  $|\psi-90^{\circ}| > 2.7^{\circ}$ , the numbers of 491 habit plane variants for each twinning type are less 492 than 48 and it is possible to obtain the habit plane 493 variants with other type of twins. What we are most 494 interested in is when  $\psi$  is in the range between 93.9° 495 and 95.1°. Then, the habit plane variants with com-496 497 pound twins are obtained. The solutions are given in Table 2 for  $\psi = 94.1^{\circ}$ . We note that for the 498  $\{001\} < 100 >$  compound twinning case there are two 499 twins with different volume fraction and each volume 500 fraction gives two sets of solutions. The correspond-501 ing transformation strain contours for  $B2 \rightarrow R$ -502

1057Table 2. The theoretical results for the habit plane normal, transform-1058ation direction and the volume fraction of twins for the compound1069twinned B19' martensite during R to B19' transformation

Solutions of {011}<100> compound twinned B19' martensite
$f = 0.18278,  \mathbf{b}_{\mathbf{B19}'}  = 0.1406$
$m_{B19'} = (-0.4080, 0.8756, -0.2585), b_{B19'} = [0.1043, 0.0273, -0.0903]$
$m_{\mathbf{B19'}} = (-0.7096, -0.2546, 0.6569), b_{\mathbf{B19'}} = [0.0647, 0.1212, 0.0300]$
$f = 0.57518,  b_{B19'}  = 0.1436$
$m_{\mathbf{B19}'} = (-0.3896, -0.3047, 0.8691), b_{\mathbf{B19}'} = [0.1090, -0.0907, 0.0225]$
$m_{B19'} = (-0.7277, 0.6503, -0.2178), b_{B19'} = [0.0637, 0.0373, 0.1232]$

Phase $\rightarrow$ B19' transformation are shown in Fig. 4(d) 503 for compression. Similar results for tension case are 504 given in Fig. 4(e). The tensile results are included as 505 a reference for comparison and point to higher trans-506 formation strains compared to compression. The 507 transformation strain levels for orientations studied in 508 the present work are listed in Table 1 for compression 509 (fourth column). The results were calculated by using 510  $\psi = 94.1^{\circ}$ . The highest transformation strain is 511 determined by selecting the maximum RSSF from all 512 possible solutions and calculating the accompanying 513 transformation strain. The results show that the trans-514 formation strains of compound twinned martensite are 515 smaller compared to the case of Type II-1 twinning 516 (compare Fig. 4(a) and (d) with Fig. 4(b) and (e)). 517

# 4. EXPERIMENTALLY DETERMINED STRESS-STRAIN RESPONSE IN THE TEMPERATURE RANGE 77–440 K 520

The stress-strain response of 51.5Ni-Ti (at small 521 strains) over a broad range of temperatures are sum-522 marized in Fig. 5(a) and (b). In these experiments the 523 specimens were strained to 3% in compression and 524 unloaded to zero stress. The [001] orientation is 525 chosen to illustrate the stress-strain results in Fig. 526 5(a) and (b) for solutionized and aged cases respect-527 ively. For the solutionized case, the pseudoleastic 528 response is observed at temperatures as low as 77 K 529 and extends to temperatures near 300 K. These results 530 point to the enormous capability of 51.5%Ni alloys 531 to exhibit transformation over a broad range of tem-532 peratures. Some degree of shape memory was 533 observed at 77 K and at higher temperatures but 80% 534 of the transformation strains was comprised of 535 pseudoleastic strains. The range of pseudoleastic 536 response is indicated on the figure. As the tempera-537 tures exceed 300 K the deformation could not be 538 recovered. When the material is overaged the marten-539 site start temperature increases to 236 K and the 540 stress-induced transformation occurs over the range 541 270-440 K. These results are illustrated in Fig. 5(b). 542 The strength levels observed in this case are lower 543 than the solutionized case, and nearly 100% pseudole-544 asticity is observed at temperatures above 270 K. 545 Similarly, the range of pseudoleasticity temperature 546 interval is marked on the figure. 547

To gain insight into the recoverable strains under 548 the application of larger strains, and to establish the 549 maximum recoverable strains, a series of incremental 550 straining experiments have been conducted. In these 551 experiments, the strain is progressively increased in 552 increments of 2%, and upon unloading the specimen 553 was heated to 373 K to realize the shape memory 554 strains. The results of the incremental straining 555 experiments for the solutionized case are given in Fig. 556 6(a)-(e) at 77 K. These experiments were conducted 557 on all five orientations considered in this study. The 558 transformation (recoverable) strains were a sum-559 mation of pseudoleastic and shape memory strains 560





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Fig. 6. Stress-strain response and the evolution of transformation strains at 77 K (a) for the [001] solutionized case, (b) for the [110] solutionized case, (c) for the [111] solutionized case, (d) for the [122] solutionized case and (e) for the [012] solutionized case.

where the shape memory component was rather
 small. This is in contrast to the behavior of <51% Ni</li>
 NiTi alloys where at low temperatures shape memory
 behavior dominates while at higher temperatures
 pseudoleastic response is most prevalent. Experiments were conducted at 77 K because at this tem-

perature the transformation strains are higher compared to higher temperatures. This was confirmed with experiments at higher temperatures. The recoverable strain levels are shown with data points in Fig. 6(a)–(e) and they increase with increasing strain, reach a maximum and then decrease with further

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could not be fully recovered upon unloading, and

heating to 100°C was required for further recovery.

The [001] orientation [Fig. 7(a)] exhibits elastic

response after austenite-martensite transformation

because plastic deformation due to slip is curtailed in

the austenitic domains. The recoverable transform-

ation strains are comparable to the solutionized case;

we note, however, that the [111] and [122] orien-

tations display significantly higher recoverable strains

(3.0% and 2.3% respectively) compared to the sol-

utionized case. The results of the experimental recov-

erable strains are listed in Table 1 (6th col.). Overall,

straining. The maximum recoverable strains are listed in Table 1 and are in the range 0.76% to 4.2%. We note that in the [111] and [122] orientations the material undergoes predominantly plastic deformation (see Fig. 1(c) for a TEM microphotograph) and the amount of recoverable strains is rather small.

The results of the incremental straining experiments at room temperature for the overaged case are given in Fig. 7(a)–(e). We note that the room temperature coincides with the austenite finish temperature, therefore, pseudoleastic response dominates at low strains. At higher strain levels the transformation

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Fig. 7. Stress-strain response and the evolution of transformation strains at 293 K (a) for the [001] overaged case, (b) for the [110] overaged case, (c) for the [111] overaged case, (d) for the [012] overaged case and (e) for the [122] overaged case

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the stress levels of the 51.5% Ni alloys are 800 MPa 597 compared to 400 MPa values for the 50.8% Ni alloy 598 studied previously [2]. 599

#### 5. DISCUSSION OF RESULTS

The results for the 51.5%Ni-Ti alloys show for the 601 first time that the temperature range over which 602 pseudoleasticity develops is as high as 200 K for the 603 case of [001] orientation [see Fig. 5(a) and (b)]. This 604 extends the utility of the NiTi alloys over a temperature range at least two times higher than the pre-606 viously reported range of 80 K [1, 18, 19]. If we can 607 tailor the texture of polycrystals to a [001] type then 608 we can develop superior compressive actuators. In the 609 case of [012] orientation, the temperature range of 610 stress-induced transformation exceeds 100 K, while 611 for the [111] case transformation does not occur due 612 to slip deformation at all temperatures studied here. 613 One major difference between the 51.5%Ni and the 614 50.8% Ni alloys is that the strength levels in 51.5% Ni 615 are higher by as much as 400 MPa over the entire 616 stress-induced transformation regime [2]. This makes 617 51.5% Ni attractive in specific applications where high 618 strength shape memory alloys are required with large 619 recoverable forces. 620

The experimental results demonstrate that the 621 recoverable (transformation) strains for the 51.5%Ni 622 composition are lower than the 50.8%Ni alloys [2]. 623 One possible explanation for the lower transformation 624 strains for the higher nickel material is the higher vol-625 ume fraction of precipitates (20%) which are untrans-626 formable. The second reason, which is explored in 627 this study, is that the transformation occurs through 628 a two step B2 to R to B19' transformation with com-629 pound twinning. The compound twinning produces a 630 lower transformation strain compared to the Type II 631 twinning case (Fig. 4(a),(b),(d) and (e)). 632

The [001] orientation has a Schmid factor for slip 633 of zero for the {001}<001>and {011}<001> sys-634 tems [20]. Therefore, slip cannot operate in this orien-635 tation. To illustrate this point the TEM results 636 obtained from the 440 K straining case are presented 637 in Fig. 8(a) and (b). Unlike other orientations [Fig. 638 1(c)], there is no indication of slip activity in the 639 [001] case. Both TEM microphotographs are obtained 640 from the same location with Fig. 8(b) displaying the 641 precipitate and austenite domains at a higher magni-642 fication. Twinning at high temperatures has been 643 observed in the [001] orientation as a main mech-644 anism of plastic deformation [20]. The pseudoleastic-645 ity temperature interval is observed for the <001>646 case as 200 K, which far exceeds the levels in 647 <111>, <122> and <112> orientations. 648

Due to the presence of the precipitates, there are 649 multiple nucleation sites for martensite formation, 650 and multiple variant formation. The number of vari-651 652 ants activated is expected to differ from the phenomenological theory. The resultant multi-variant struc-653 ture produces transformation strains that exhibit 654





100 nm

Fig. 8. (a) The precipitate structure and the surrounding matrix for the [001] compression experiment under straining at 440K. (b) The same location as (a) but a higher magnification demonstrating that slip activity is indeed absent for this orientation.

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smaller orientation dependence compared to the pre-655 vious work on 50.8% Ni crystals. For example, for 656 the 51.5% Ni alloy, the transformation strains are in 657 the range 2.3–3.7% for the overaged case (Table 1). 658 The variation in the transformation strains for the 659 50.8% material was in the range 2.5-5.6% [2]. We 660 also note the absence of the plateau regions in the 661 stress-strain curves (Fig. 6(a),(b) and (e) and Fig. 662 7(a)-(e)), which are characteristic of the transform-663

ation boundary motion of a single variant. The pres-664 ence of the precipitates also influences the back stress, 665 which governs the reverse transformation stress and 666 the overall hysteresis stress range. With increasing 667 strain in Fig. 7(a)-(e), the reverse transformation 668 stress level decreases. Ultimately, the reverse trans-669 formation stress nears zero stress for favorable orien-670 tations upon unloading from large strains. 671

For the NiTi alloys, it has been widely recognized 672 that <011> Type II twinning is the most prevalent 673 twinning mechanism in B19' martensites. There are, 674 however, a number of investigations [11, 12, 17] 675 where (001) compound twinning has been reported. 676 Because the (001) compound twinning is not a lattice 677 invariant shear, it has been viewed as deformation 678 twinning by a number of investigators. We demon-679 strated in this study that compound twinning is indeed 680 a lattice invariant shear if the transformation to B19' 681 is achieved via a two step B2 to R to B19' transform-682 ation. The first study that analyzed such a two step 683 transformation (B2 to R to B19') was reported by 684 Krishnan [12]. However, we note that the compound 685 twinning solution presented in this paper is different 686 from Krishnan's in several respects. We provide a 687 solution for major and minor twin volume fractions 688 within the B19' phase and is not restricted to the 689 f = 0.5 assumption made by Krishnan [12]. Also, we 690 treat the R-phase as internally twinned as opposed to 691 the single crystal R phase assumed by Krishnan [12]. 692 The results confirm that the transformation strains for 693 the compound twinning case [Fig. 4(a),(b),(d) and (e)] 694 are lower than the Type II-1 twinning results. 695

Our previous work has shown that the end of the 696 stress plateau is not the end of the stress-induced 697 transformation [2]. During austenite to martensite 698 transformation large regions of undeformed austenite 699 exists which undergoes slip deformation [2]. If the 700 slip deformation in the austenite regions is curtailed 701 via crystal orientation or different processing routes, 702 this would increase the recoverable strains consider-703 ably. The calculations for the Schmid factors of the 704 austenite (Table 3) pointed out the lowest Schmid fac-705 tors near the [001] pole resulting in transformation 706 707 (Fig. 7(a)) under elastic conditions. Austenite yielding is not expected in this orientation due to the prevail-708 ing {001}<001> and {011}<001> slip systems [2, 709 20]. The critical stress versus temperature results 710 given in Figs 2 and 3 confirm the remarkable resist-711 ance to slip in the [001] direction. In particular, in 712

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Table 3. Calculated Schmid factors for the austenite slip systems

Crystal orientation	Schmid factor— austenite {001}<001>	Schmid factor— austenite {011}<001>
[001]	0.00	0.00
[110]	0.50	0.50
[111]	0.33	0.47
[012]	0.40	0.28
[122]	0.33	0.47

both the solutionized and overaged cases, the increase 713 in flow stress with increasing temperatures in excess 714 of 400 K is remarkable. 715

We note that the critical stress for martensite yield-716 ing is rather high and is of the order of 2000 MPa 717 while the austenite yield levels are of the order of 718 1200 MPa. The austenite yield stress at room tem-719 perature can be established by extrapolating the criti-720 cal stress versus temperature slope (beyond  $M_d$ ) to 721 lower temperatures. This is easier to demonstrate for 722 the case of overaged material (Fig. 3). Ultimately, 723 when martensite yielding occurs then the recoverable 724 strains decrease with further increase in applied strain 725 (see Fig. 6(a) and Fig. 7(b),(d) and (e) for a clear 726 display). Therefore, any treatments that increase the 727 critical austenite stress, such as through increase in 728 volume fraction of precipitates, or texture in the case 729 of polycrystals would increase the transformation 730 strain. 731

Another important observation is that the  $M_d$  tem-732 perature is nearly 360 K for the overaged crystals in 733 all the crystallographic orientations except the [001] 734 case where the  $M_d$  temperature is much higher. Since 735 the  $M_{\rm s}$  temperature has been measured as 236 K for 736 the overaged case the range of transformation is at 737 least 120 K for most orientations while this range is 738 in excess of 200 K in [001] orientation. Also, the 739 strength increases with increasing temperature in the 740 range 200-360 K for the solutionized case for all 741 crystal orientations. In the [001] case the increase in 742 strength occurs well past 360 K. The results show 743 that the deformation temperature relative to the  $M_{\rm d}$ 744 temperature is an important consideration in interpret-745 ation of the stress-strain response. 746

# 6. CONCLUSIONS

- 1. The strength of the 51.5Ni-Ti alloys is substan-749 tially higher (nearly 400 MPa) than the 50.8%Ni 751 NiTi alloys both in the martensitic and the austen-752 itic states. The strength of the [001] orientation 753 increases with increasing temperature exhibiting 754 pseudoleasticity over a broad range of tempera-755 tures. This orientation does not display an  $M_d$  tem-756 perature up to 440 K. In other orientations the 757 pseudoleasticity is substantially lower compared to 758 the [001] case. The results confirm that [001] 759 orientation can be exploited for specific appli-760 cations. 761
- 2. The strength of the solutionized alloy is substan-763 tially higher than the overaged case at low tem-764 peratures. The solutionized condition exhibits 765 higher transformation strains compared to the 766 overaged case in [001] and [011] (~3.9 and 4.2% 767 respectively) but much lower transformation 768 strains in [111] and [122] orientations (<1%). Slip 769 deformation is forwarded as an explanation for the 770

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very low transformation strains in the [111] and 771 [122] directions. 772 3. The overaged specimens exhibit pseudoleasticity 773 and shape memory at temperatures above 273 K. 775 The experimental transformation strain levels are 77e 48(13), 3311. substantially lower than the theoretical calcu-777 lations in compression. Two factors were sug-778 gested to explain the lower transformation strains 779 observed. These include the inability of the pre-34. 2045. 780 cipitates to transform and the compound twinning 781 solution producing lower transformation strains. 782 4. A new model for transformation was introduced 783 with compound twinning as a lattice invariant 785 shear solution. A solutions was obtained for the 786 36(1), 181. two stage  $B2 \rightarrow R \rightarrow B19'$  transformation. In the 787 development of this model, the R-phase is intern-788 1988, 57(3), 467. ally twinned and no assumptions were made 789 regarding the twin volume fractions in the B19' 790 phase. The results point out that depending on the 791 crystal orientation and the loading direction, the 792

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pared to the Type II-1 twinning (single step transformation) in compression. 796 Acknowledgements-Portions of the research is supported by 798 a grant from the National Science Foundation contract CMS 799 99-00090, Mechanics and Materials Program, Arlington, Vir-800 ginia, and Air Force Office of Scientific Research, Directorate 801 802 of Aerospace and Materials Sciences, Arlington, Virginia. Professor Chumlyakov received support from the Russian Fund 803 for Basic Researches, Grant Nos. 02-95-00350, 99-03-32579. 804

two stage transformation produces compound

twinning with lower transformation strains com-

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